



An Update on the Charged Aerosol Release Experiment *CARE*



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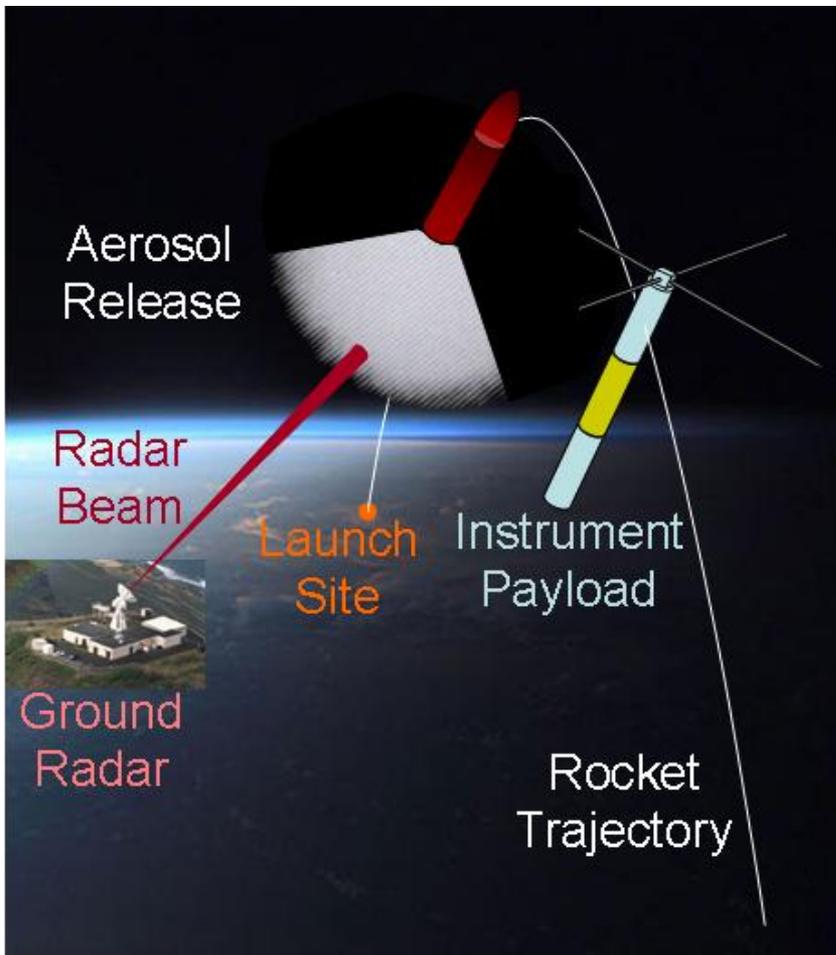


12th Workshop on the
Physics of Dusty Plasma
19 May 2009





Charged Aerosol Release Experiment Modified/Enhanced *CARE* Concept



Radar and In Situ Diagnostics of Artificial Dusty Plasma Cloud

Objectives: Examine the effect of artificially-created, charged-particulate layers on the penetration of UHF, L-Band and S-Band radars.

Design and build a rocket payload with a chemical release and instruments to solve the mystery of radar echoes from dusty plasmas

Description: A series of rocket experiments will be conducted to test the theories for radar scatter from charged dust.

– Rocket Experiment

- Chemical Release (280 km Altitude, 111 kg)
- Instrumented Payload
 - Radio Beacon (CARE I)
- Ground Radars for Backscatter Observations
- Launch Sites
 - Wallops Island, Virginia (CARE I 2009)
 - Andoya, Norway or Poker Flat, Alaska (CARE II)
 - Kwajalein, Marshall Islands (CARE II Alternate)

– Supporting Theory

- Chemical Release: Plume Model
- Particle Dispersal: Monte Carlo Simulations
- Plasma Turbulence: Electrostatic PIC Code
- Radar Scatter: Electromagnetic Theory



CARE Personnel



<u>Name</u>	<u>Affiliation</u>	<u>Name</u>	<u>Affiliation</u>
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Prof. Wayne Scales	Virginia Tech	Eric Johnson	NSROC/EE-Power
Prof. Marlene Rosenberg	UC San Diego	Valerie Gsell	NSROC/GNC
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Dr. Phil Erickson	Millstone Hill	Ron Walsh	NASA/PM
Dr. George Jumper	AFRL/RVBYA	Libby West	NASA/SRPO

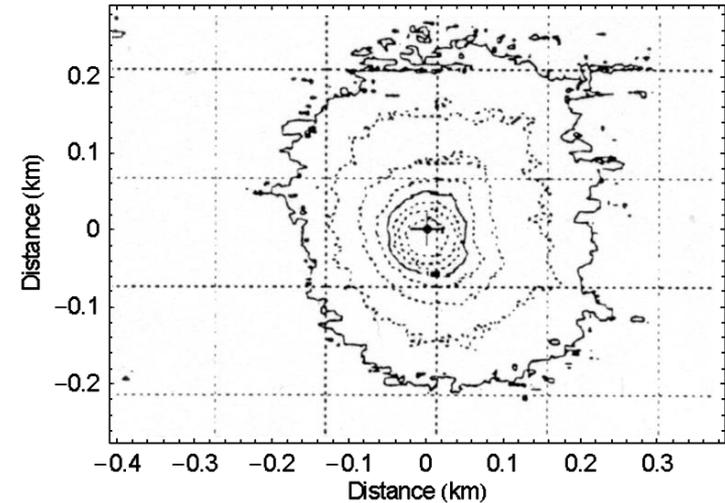


Particulate Release Observations During the SCIFER Mission on 25 January 1995

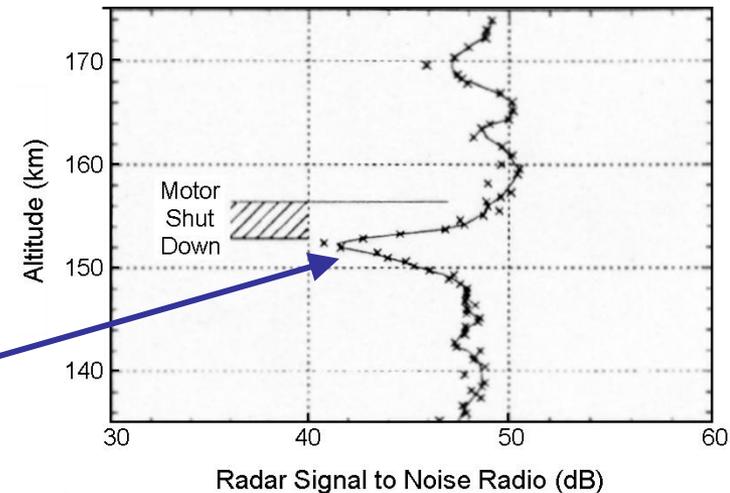


- 25 January 1995 Four Stage Black Brandt XII Launch from Andoya Norway
- Nihka Motor Fired from 98.7 to 156.3 km Altitude
- Dust Cloud Observed for 66 Minutes
- 5.6 GHz Radar Echoes Attenuated by 7 dB

DLR Radar Video Image: SCIFER 25 Jan 1995, 151.9 km Altitude

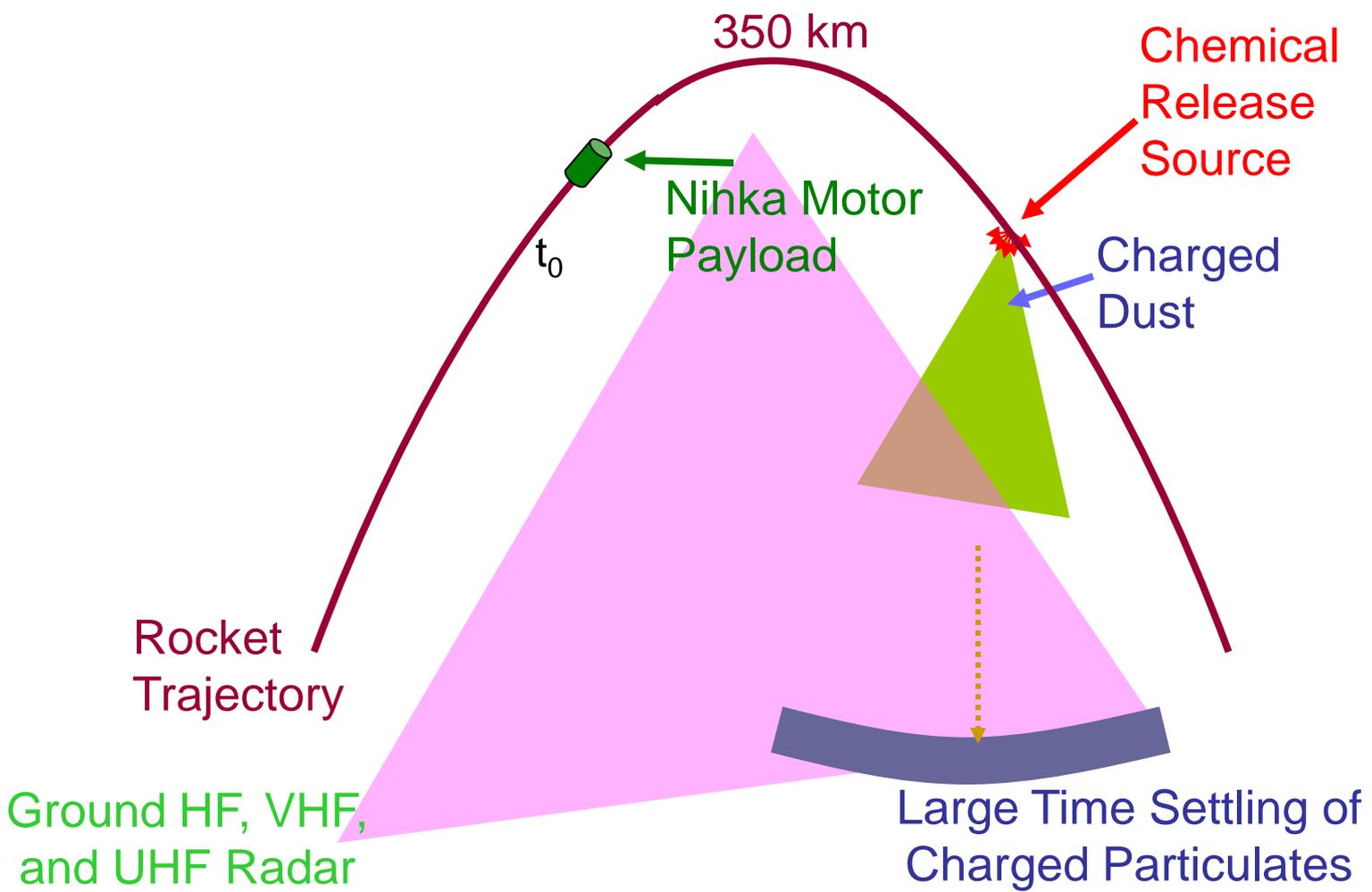


5.7 GHz DLR-Radar, SCIFER, 27 Jan 1995 06:26 UT





New Artificial Dust Layer Concept





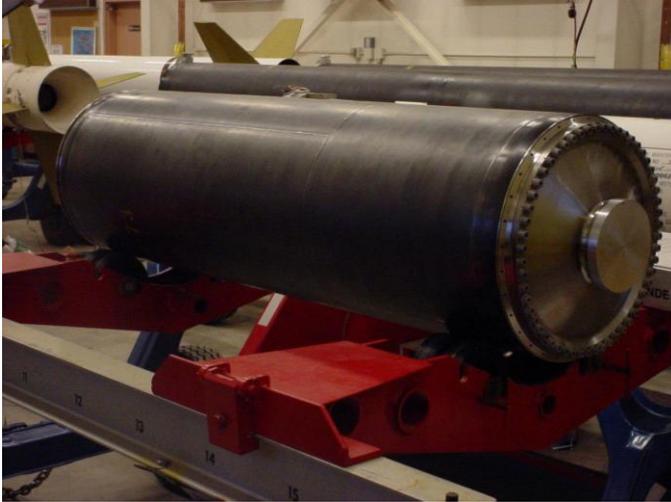
Objectives of CARE Program



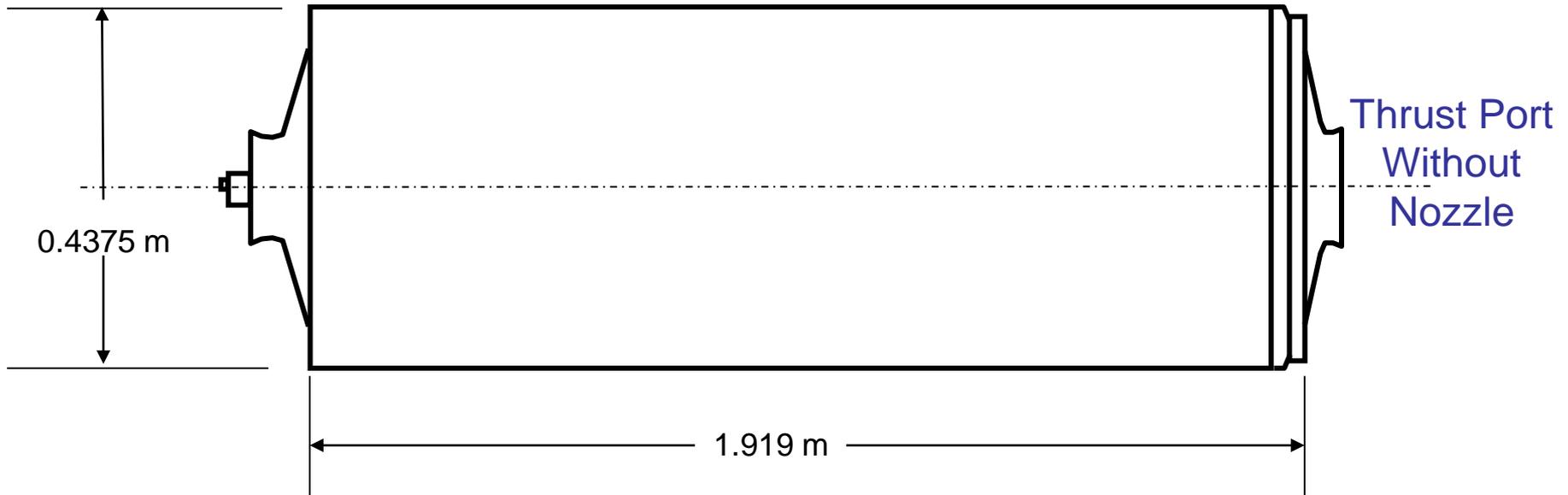
- Artificial Noctilucent Cloud Formation
 - Physics of Enhanced Radar Scatter
 - Radar, Lidar and Optical Diagnostics
 - Satellite Measurements (AIM)
- Release from Nihka Solid Rocket Motor
 - Large Concentration of Dust
 - Supersonic Injection Velocity
- Experiment Enhancements
 - Ground and Ship Ionosonde Diagnostics
 - Direct Injection by Chemical Release Module



Nihka Motor Dust Generator for CARE (111 kg Al_2O_3 in 17 Seconds)

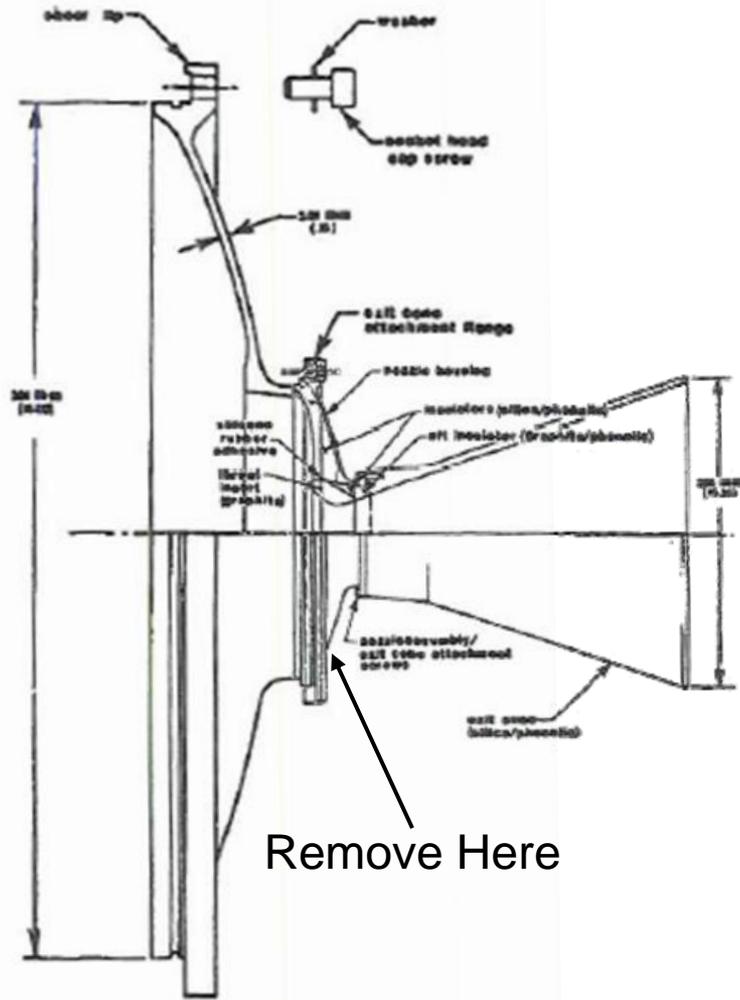


Constituent	Mole Fraction	Total Mass
Carbon Monoxide	0.2039	69.1 kg
Carbon Dioxide	0.0186	9.9 kg
Monatomic Chlorine	0.0124	5.3 kg
Hydrogen Chloride	0.1478	65.2 kg
Hydrogen	0.2712	6.6 kg
Monatomic Hydrogen	0.0317	0.4 kg
Water	0.1393	30.4 kg
Nitrogen	0.0817	27.7 kg
Aluminum Oxide	0.0897	110.6 kg



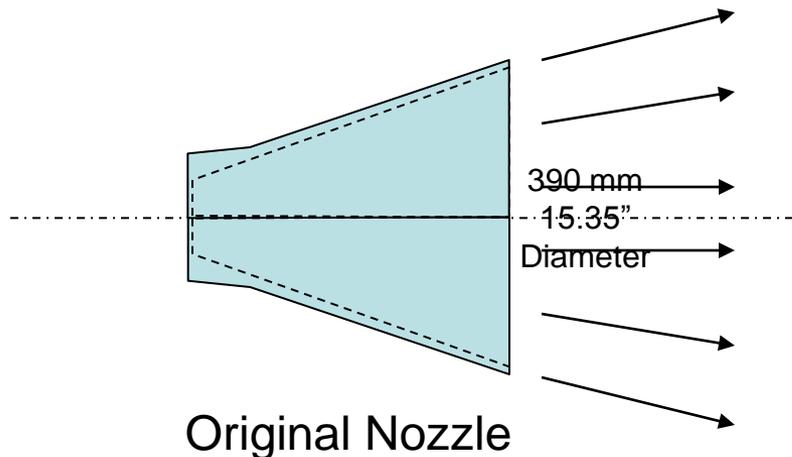
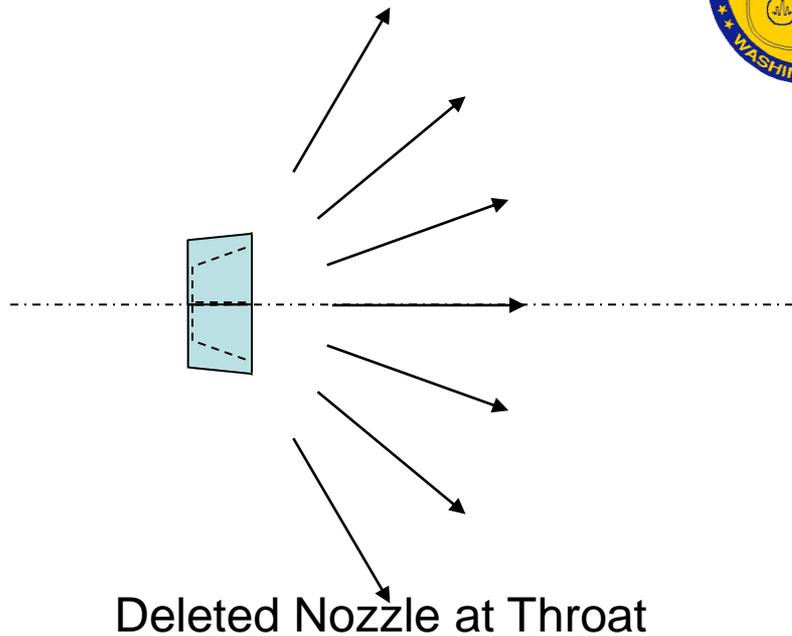


Truncated Nozzle for Nihka Motor



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FIG. 3 NOZZLE HOUSING





Michigan**Engineering**

DSMC Computations of Nozzle Effect on Plume Properties for Nihka Motor

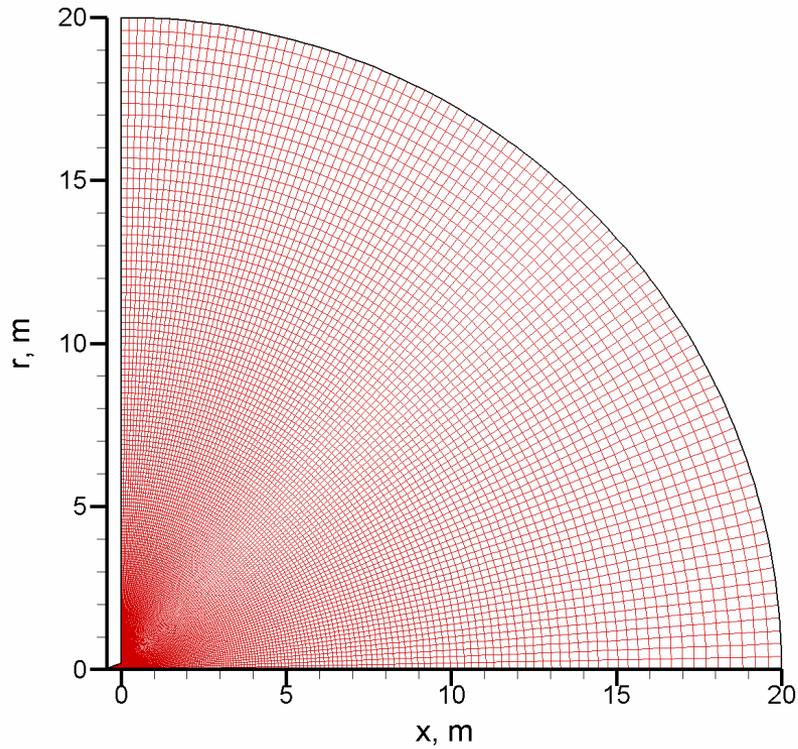
Jonathan Burt and Iain Boyd

May 2009

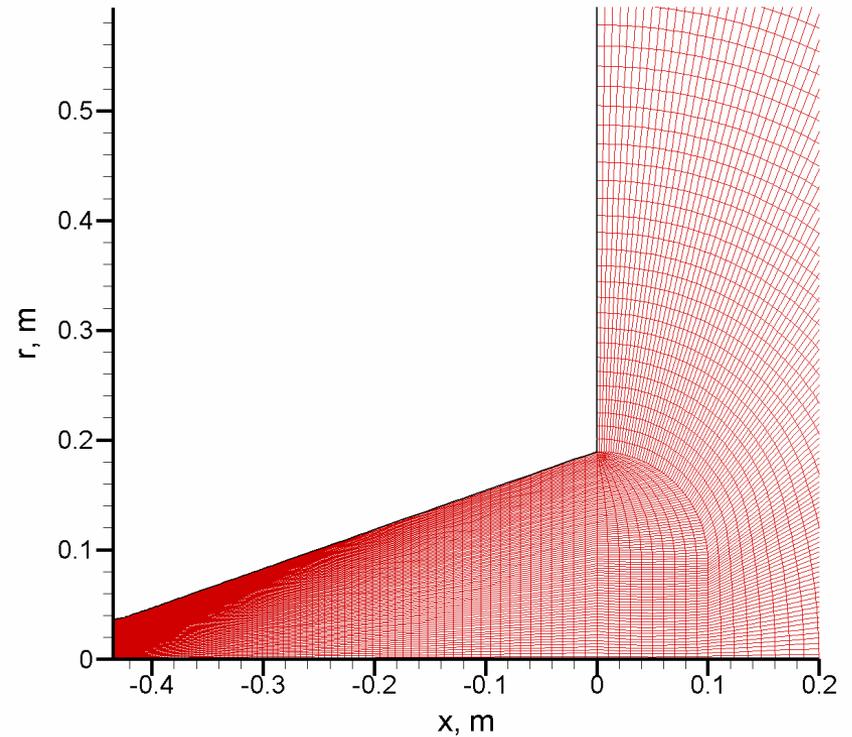


With Nozzle

- Computational mesh



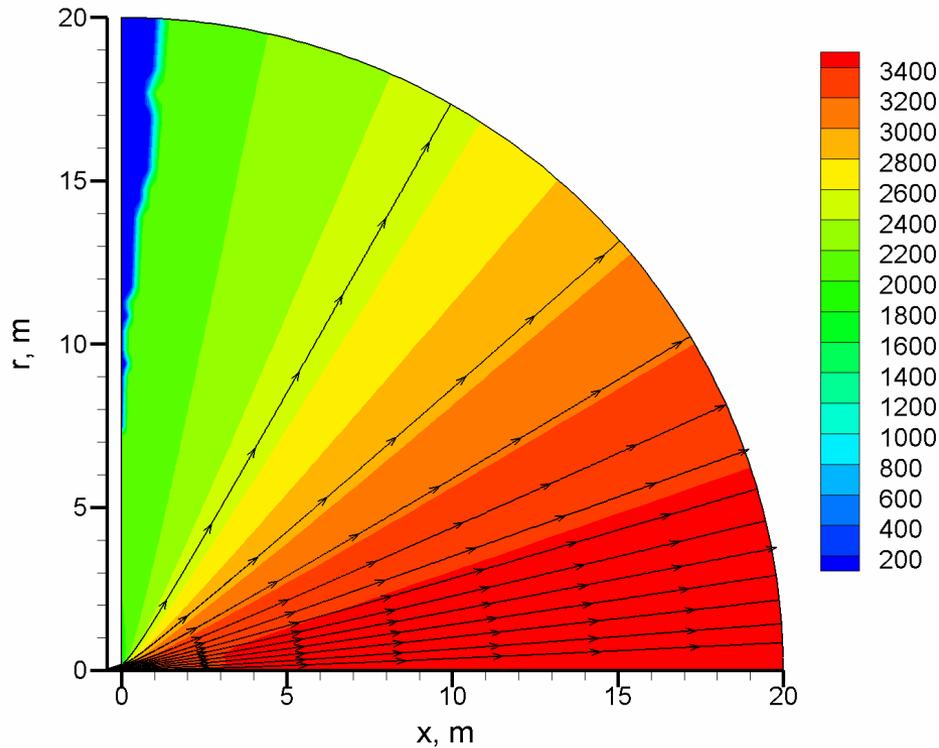
Flow Field



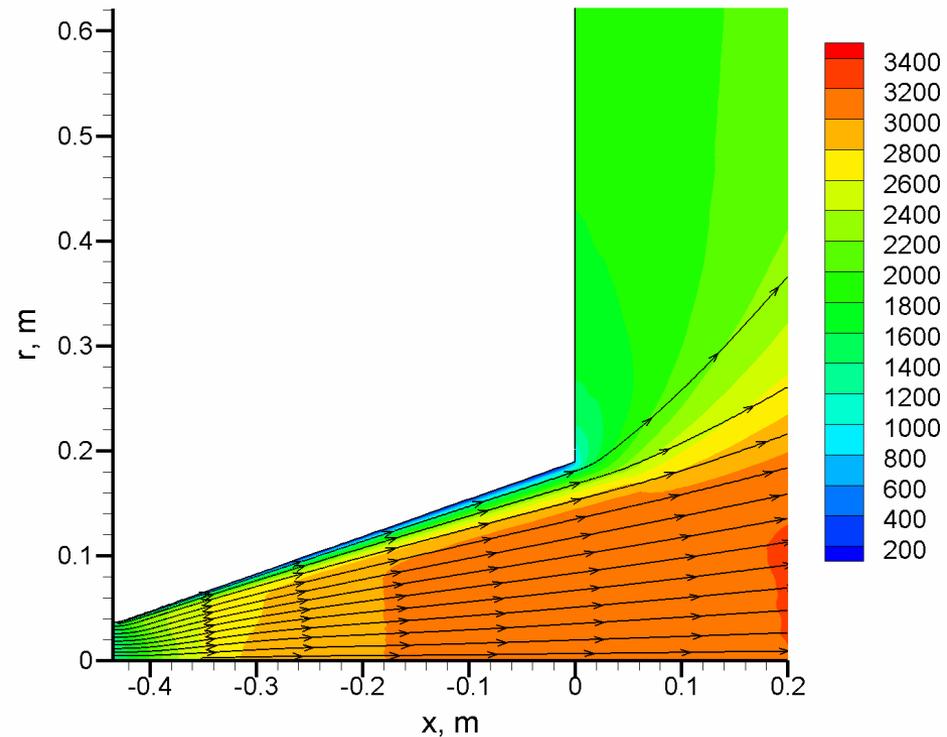
Detail at Nozzle

With Nozzle

- Streamlines, contours of bulk gas velocity (m/s)



Flow Field

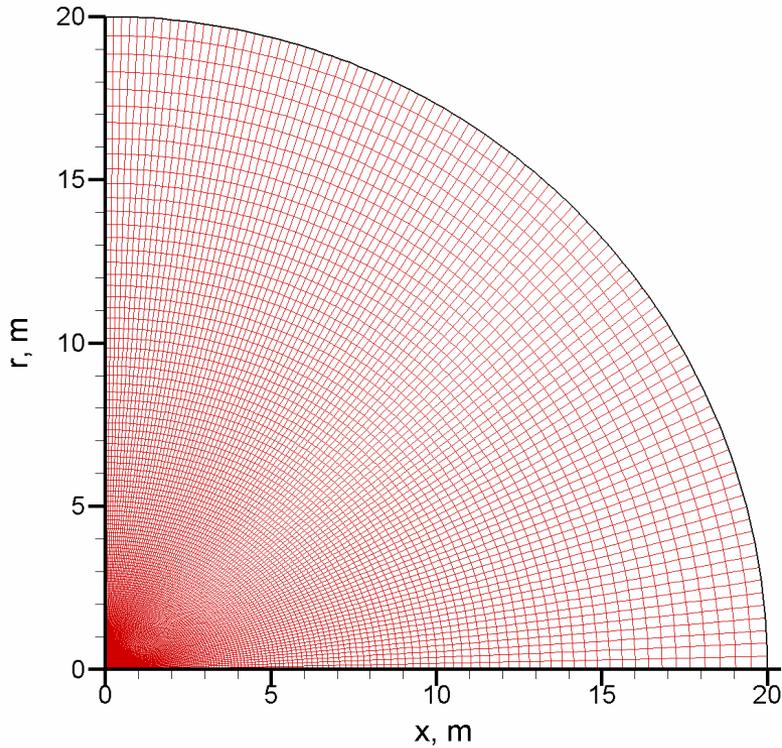


Detail at Nozzle

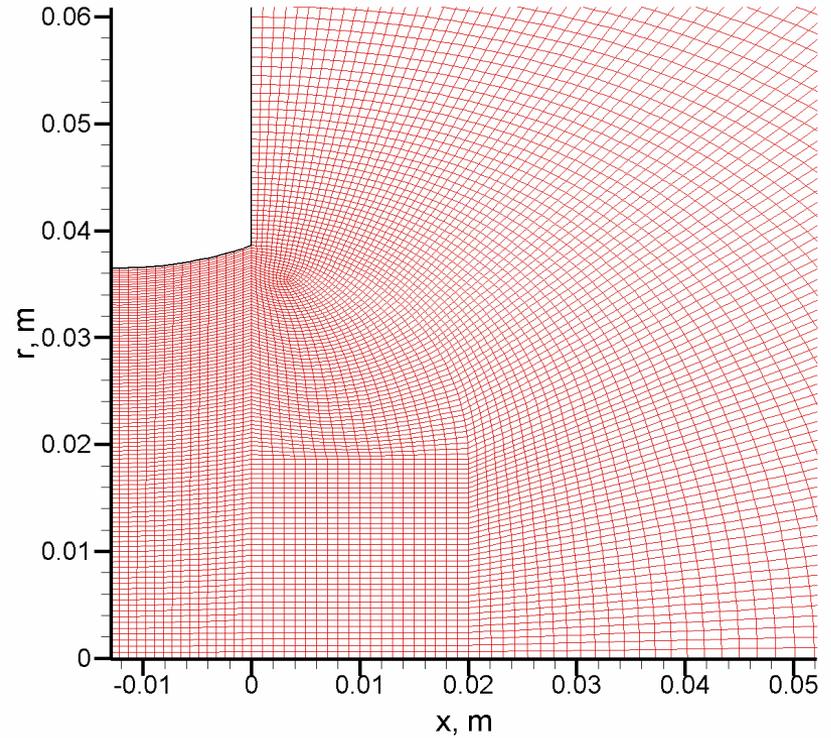
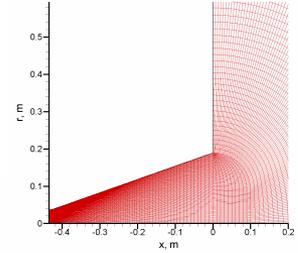


Nozzle Removed

- Computational mesh



Nozzle mesh for comparison

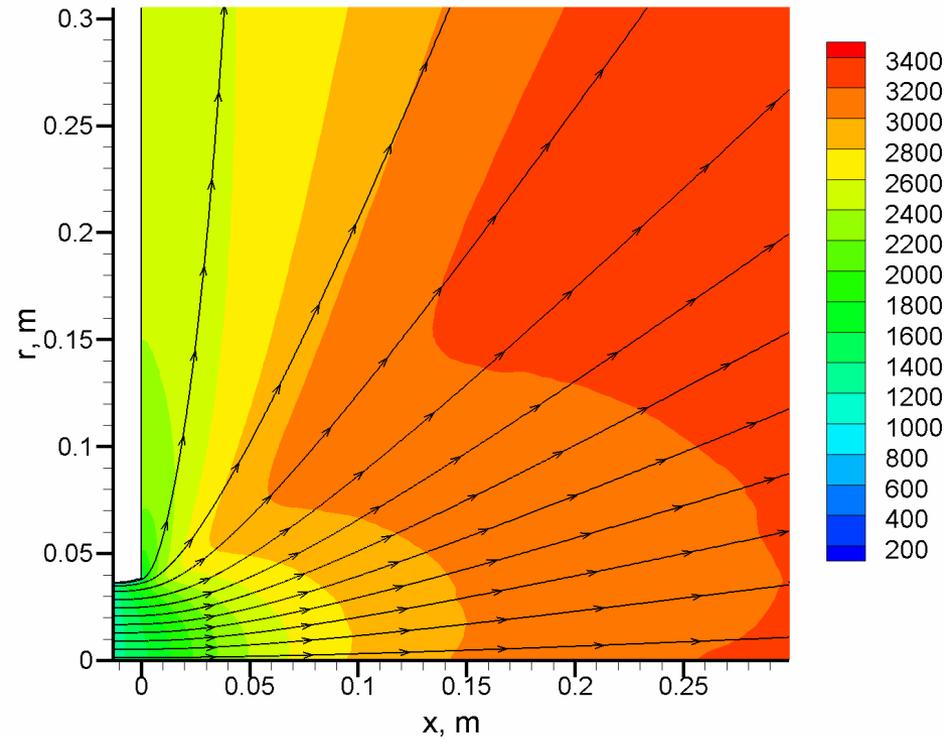
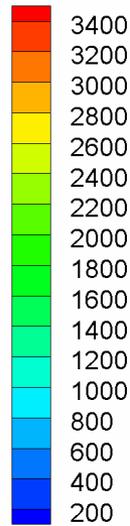
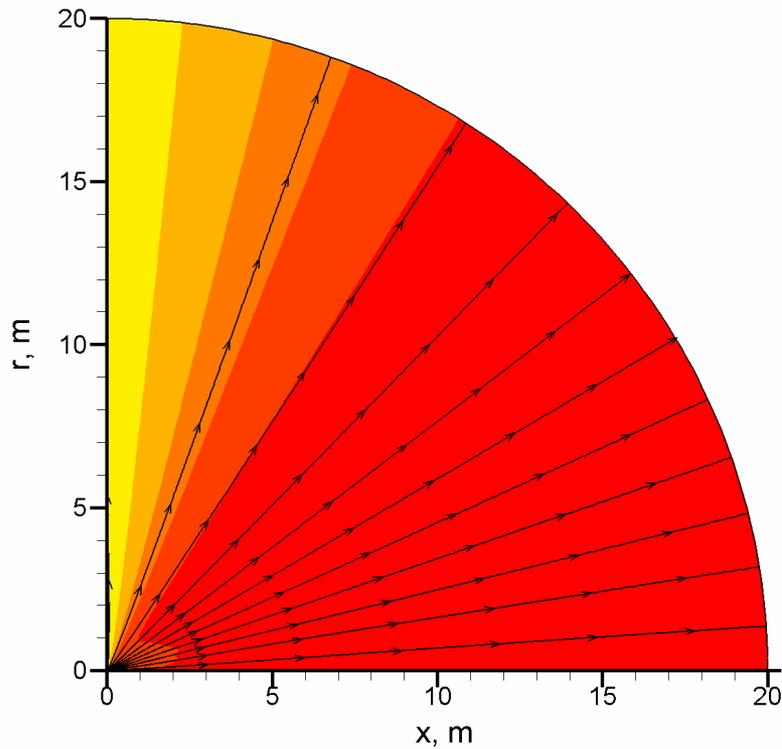


- Nozzle is truncated 1.3 cm downstream of the throat



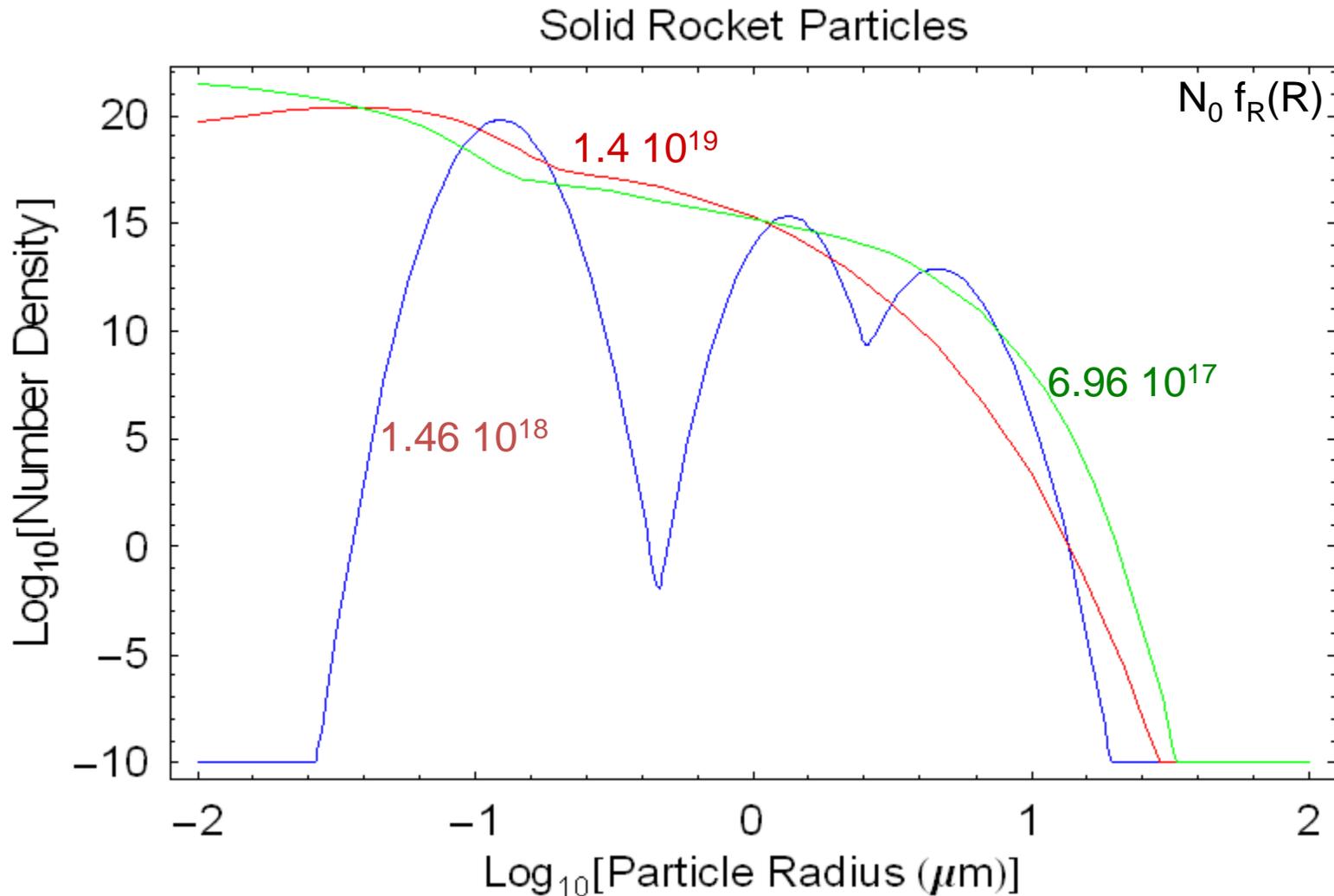
Nozzle Removed

- Streamlines, contours of bulk velocity (m/s)



Distributions of Solid Rocket Particulates

- (1) Mueller, A. C. and D. J. Kessler, The Effects of Particulates from Solid Rocket Motors Fired in Space, . Adv. in Space Res., 5, 77-86, 1985.
- (2) Jackman, CH, DB Considine, and EL Fleming, A Global Modeling Study of Solid Rocket Aluminum Oxide Emission Effects on Stratospheric Ozone, Geophys. Res. Lett., 25, 907-910, 1998.
- (3) S Gossé, L Hespel, P Gosart and A Delfour, Morphological Characterization and Particle Sizing of Alumina Particles in a Solid Rocket Motor, J. Prop. Pwr., 22, 127-135, 2006.

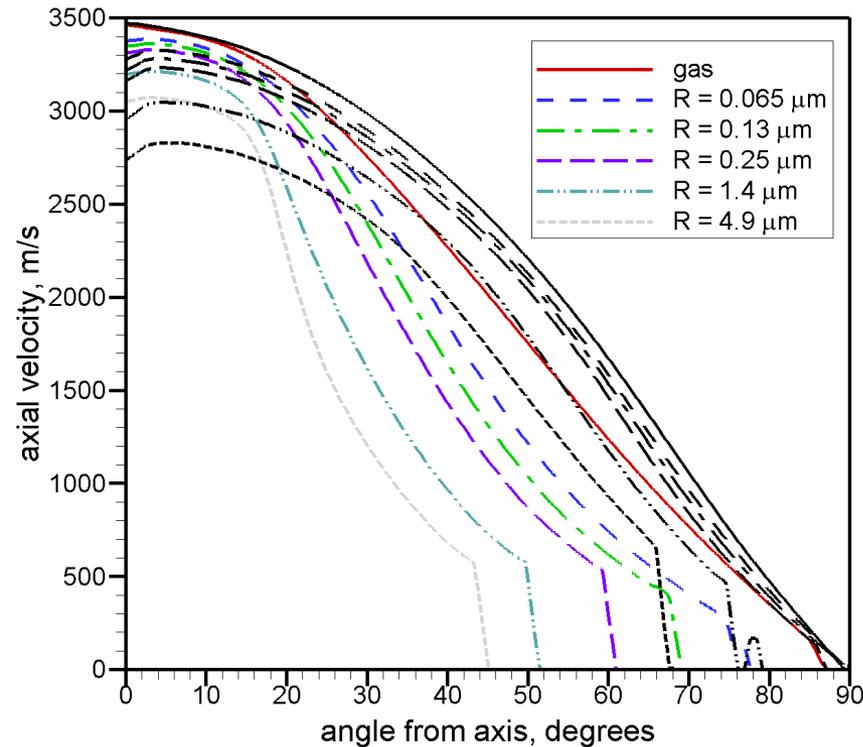




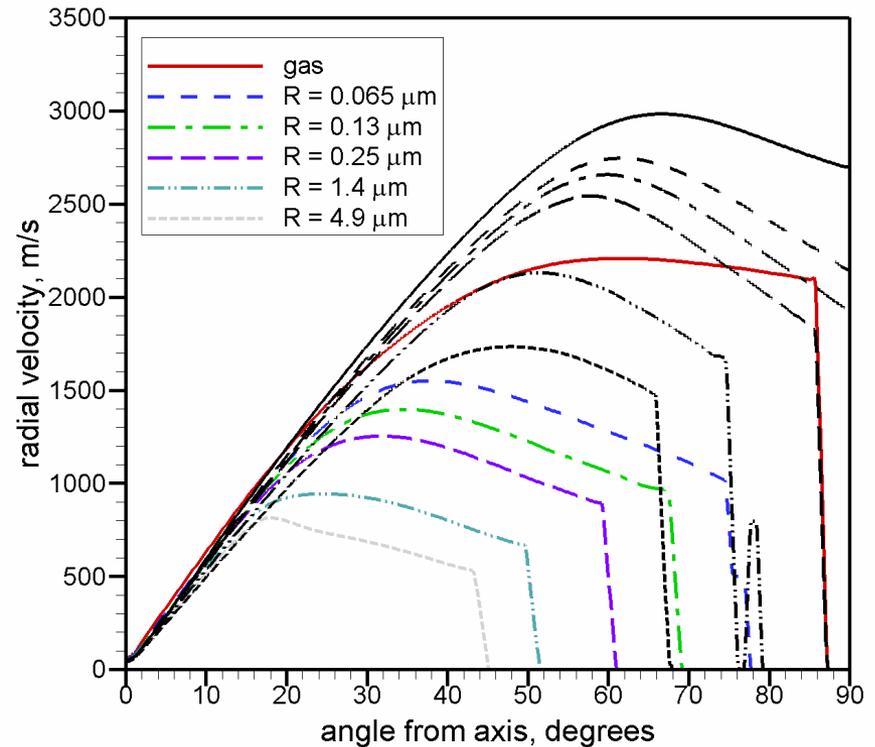
Farfield Comparison

- Comparison between **nozzle (multicolored lines)** and without nozzle (**black lines**) cases for axial and radial velocity along outflow boundary

Axial Velocity (m/s)



Radial Velocity (m/s)





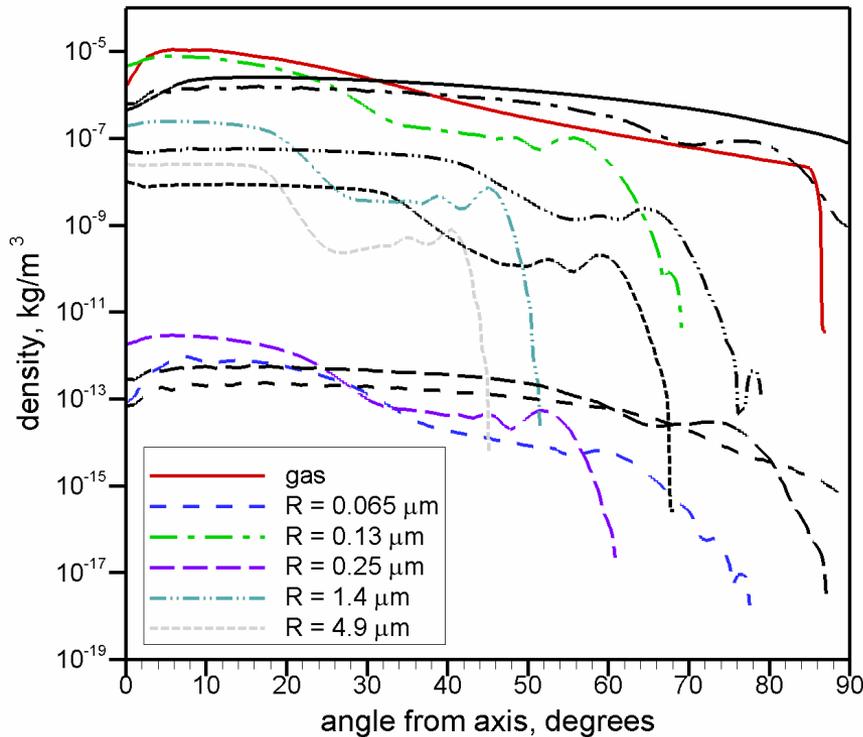
Farfield Comparison



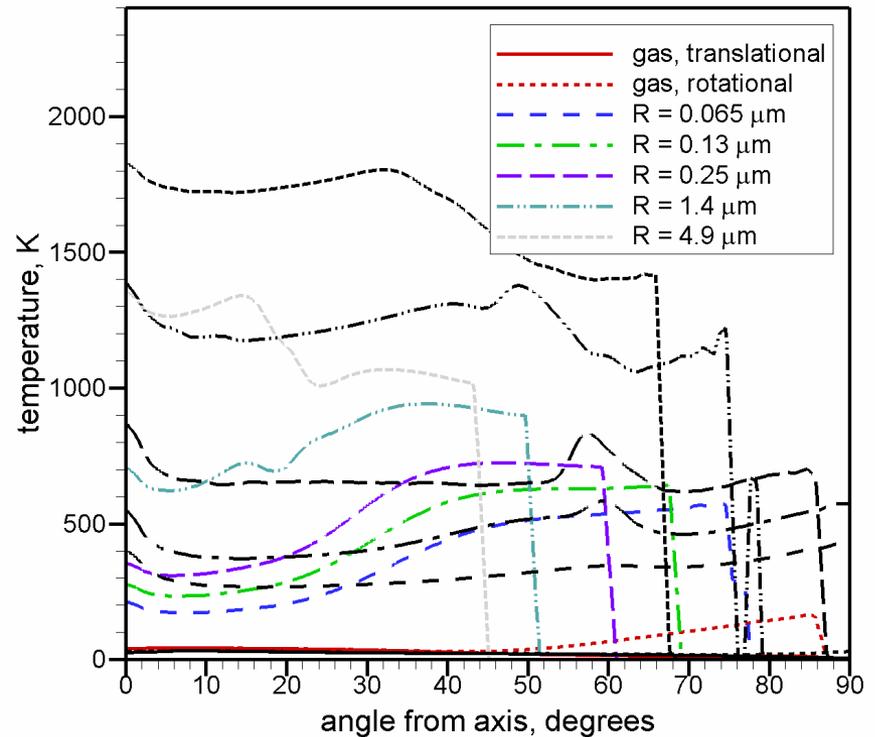
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- Comparison between **nozzle (multicolored lines)** and without nozzle (black lines) cases for density and temperature along outflow boundary

Density (kg/m^3)



Temperature (K)





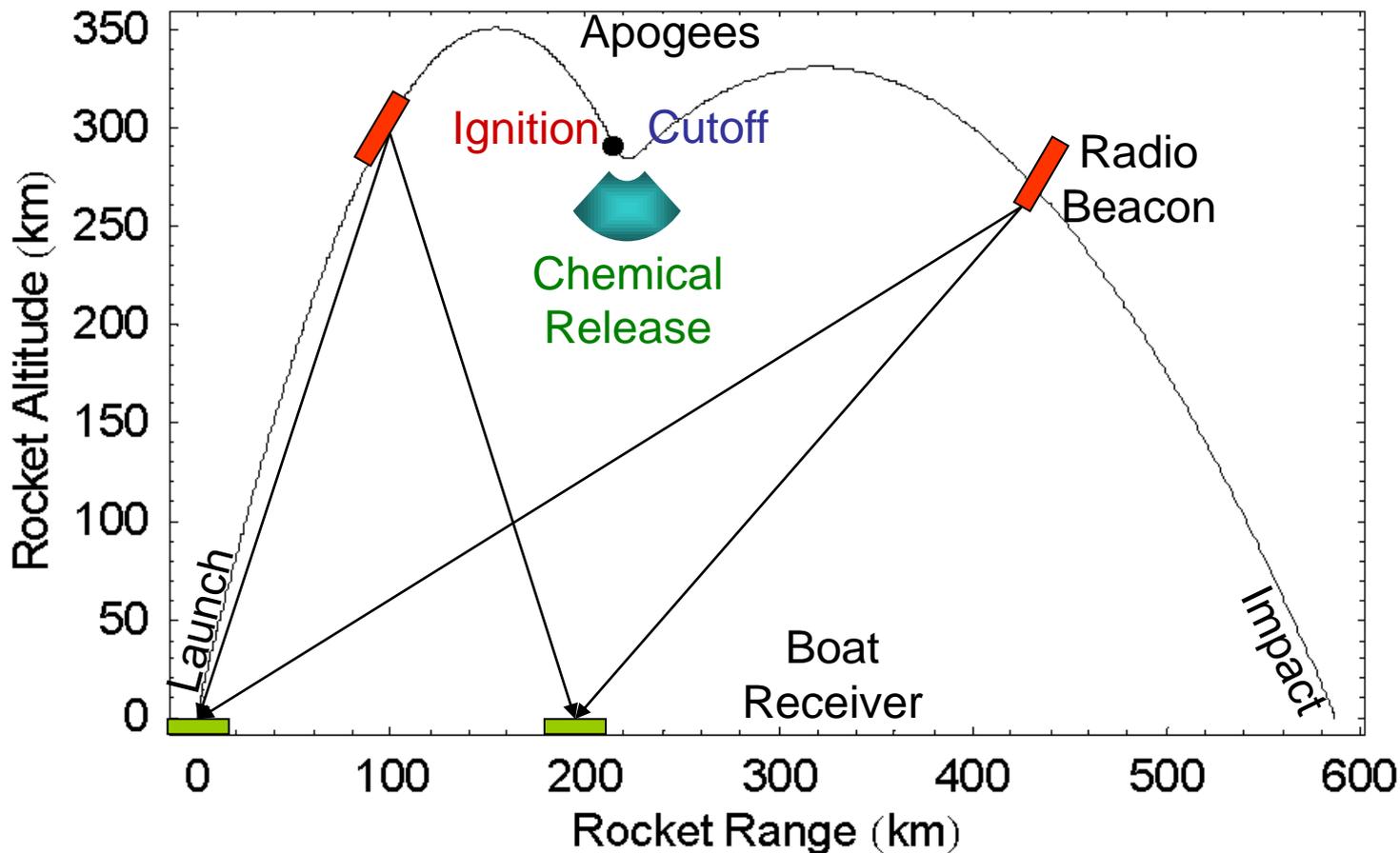
Thrust Calculation

- Thrust is evaluated by integrating contributions due to pressure and axial momentum flux from the gas and from each particle size over the farfield outflow boundary.

Case:	Nozzle	Without nozzle
Gas contribution (kN)	22.58	13.76
Particle contribution (kN)	11.75	7.40
Total (kN)	34.33	21.17



Nozzle Nihka Motor Chemical Payload Trajectory and Radio Propagation Experiment for CARE





Dynamic Dust Model for CARE



- Dust Charging (Charge Impact, Photo-Detachment)

$$Q_d = 4\pi\epsilon_0 r_d \phi_d, \quad \frac{dQ_d}{dt} = I_e + I_i + I_p, \quad I_e = \sqrt{8\pi} r_d^2 q_e n_e v_{Te} \exp\left[-\frac{q_e \phi_d}{KT_e}\right]$$

$$I_i = \sqrt{8\pi} r_d^2 q_i n_i \sqrt{v_{Ti}^2 + v_T^2 + v_r^2} \left[1 - \frac{q_i \phi_d}{KT_i}\right], \quad I_p = -\pi r_d^2 q_e J_p Q_{ab} Y_p \exp\left[\frac{q_e \phi_d}{KT_p}\right]$$

- Dust Particle Equation of Motion (Drag, Lorentz Force, Electric Fields, Gravity, Mean-Free-Path)

$$\frac{d^2 \mathbf{r}(t)}{dt^2} = -\nu_a(\mathbf{r}, \mathbf{u}) \frac{d\mathbf{r}(t)}{dt} + \frac{Q_d(t)}{m_d} (\mathbf{E} + \frac{d\mathbf{r}}{dt} \times \mathbf{B}) + \mathbf{g}$$

$$\nu_a(r, u) = \frac{\sqrt{V_r^2 + V_T^2}}{L_p(r)}, \quad L_p(r) = \frac{2 m_p}{C_D \rho_n(r) A_p} = \frac{8 \rho_P}{3 \rho_n} r_d$$

$$\text{with } A_p = \pi r_p^2, \quad m_p = \rho_P \frac{4}{3} \pi r_p^3, \quad C_D = 2 \text{ for } r_d \ll L_p$$



Dust Electric Fields

- Quasi-Neutrality

$$q_i n_i + q_e n_e + n_d \langle Q_d \rangle = 0; n_e = n_i - n_d \frac{\langle Q_d \rangle}{q_e}$$

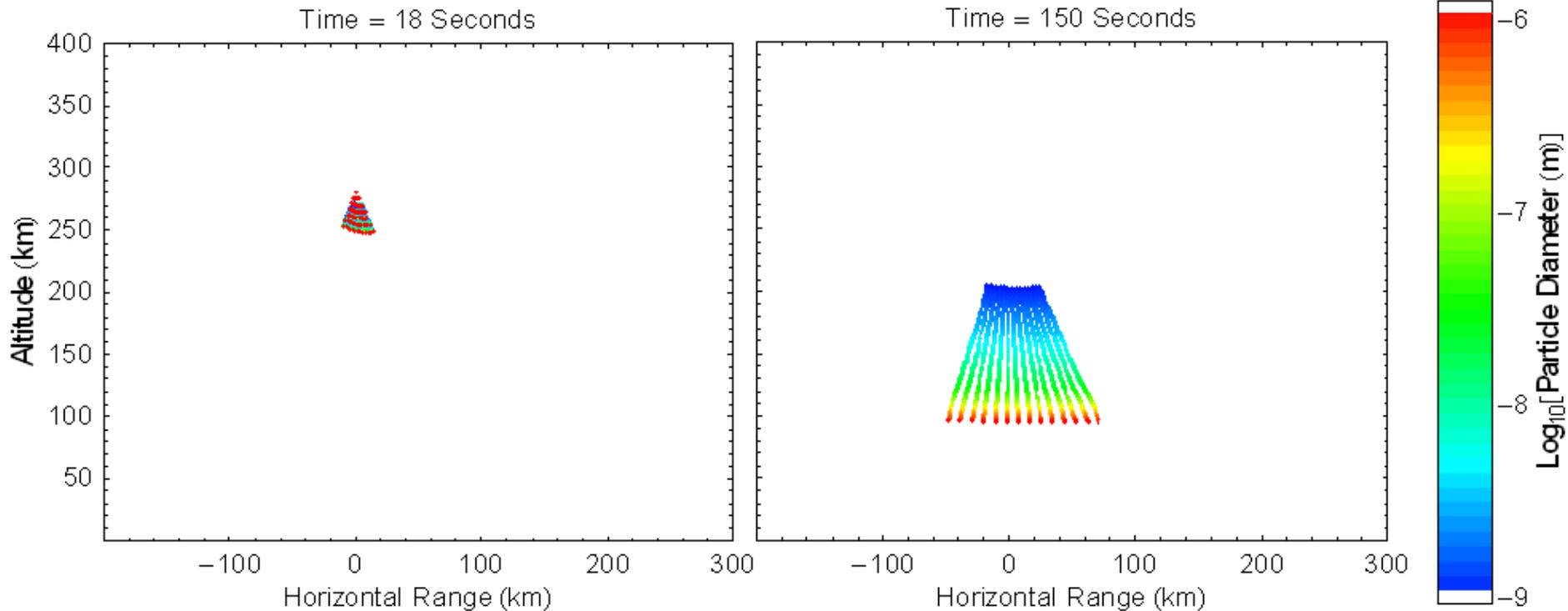
- Ambipolar Electric Field

$$E_{\parallel} = \frac{kT_e \nabla_{\parallel} n_e(z, t)}{q_e n_e(z, t)} = \frac{kT_e \nabla_{\parallel} (q_e n_i - n_d \langle Q_d \rangle)}{q_e (q_e n_i - n_d \langle Q_d \rangle)} \approx - \frac{kT_e \nabla_{\parallel} (n_d \langle Q_d \rangle)}{q_e (q_e n_i - n_d \langle Q_d \rangle)}$$

- Transient Parallel Electric Fields
- Electron Acceleration?
- Effect on Ions?



20 Degree Nozzle Dust Layer





Radar Observation Geometry of Rocket Launches from Wallops Island by the Millstone Hill Radar



Millstone Hill Incoherent Scatter Radar at MIT Haystack, MA

- 440 MHz UHF Radar
- Fully Steerable 150 foot Dish
- Line of Sight Densities, Drifts, and Temperatures



CARE Ground Radars and Lidars



- HF Radar
 - Wallops Ionosonde (Digisonde and Dynasonde)
 - SuperDARN (Virginia Tech)
 - Dr. Raymond A. Greenwald, Ruohoniemi Mike,
 - Joseph Baker
 - Data Products
 - HF Backscatter
 - Background Ionosphere
- VHF Radar
 - 50 MHz Penn State/U. Illinois
 - Professor Julio Urbina
 - Data Product
 - Background Irregularities
 - VHF Backscatter
- UHF Radar
 - 440 MHz Millstone Hill Radar, Haystack, MA
 - Phil Erickson, Frank Lind, John Foster
 - Data Products
 - UHF Backscatter
 - Background Ionosphere
- LIDAR (Laser and Telescope)
 - NRL Optical Test Facility, MRC, VA
 - 0.5, 1.0, 1.5 micron Lasers with 5 ps Pulse
 - Linda Thomas and Ray Burris
 - Data Products
 - Particle Size Distribution
 - Transport of Dust



NASA Wallops
Dynasonde Antenna



MIT Millstone Hill Radar



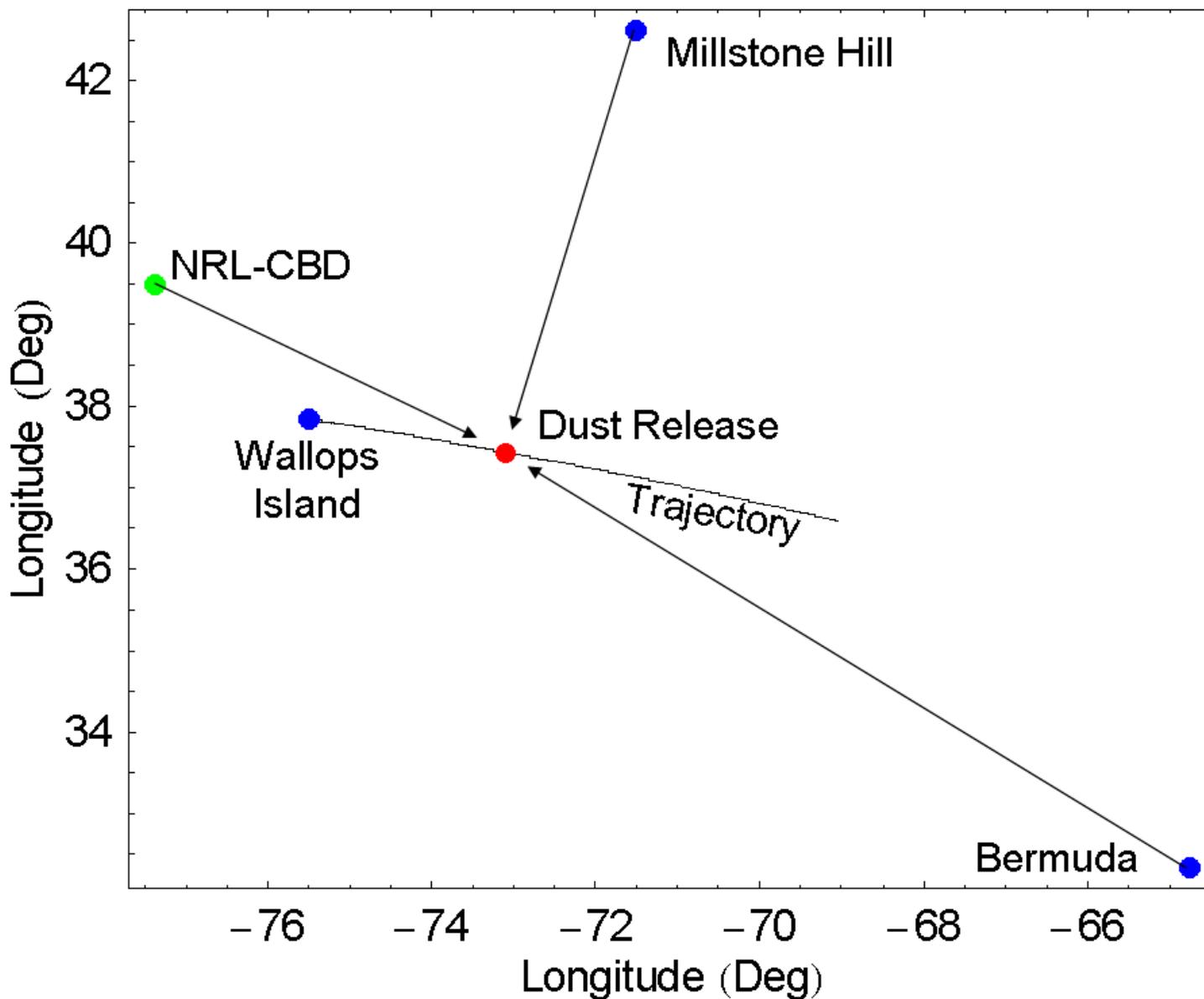
NRL Optical Test Facility



Ground Diagnostic Sites for CARE

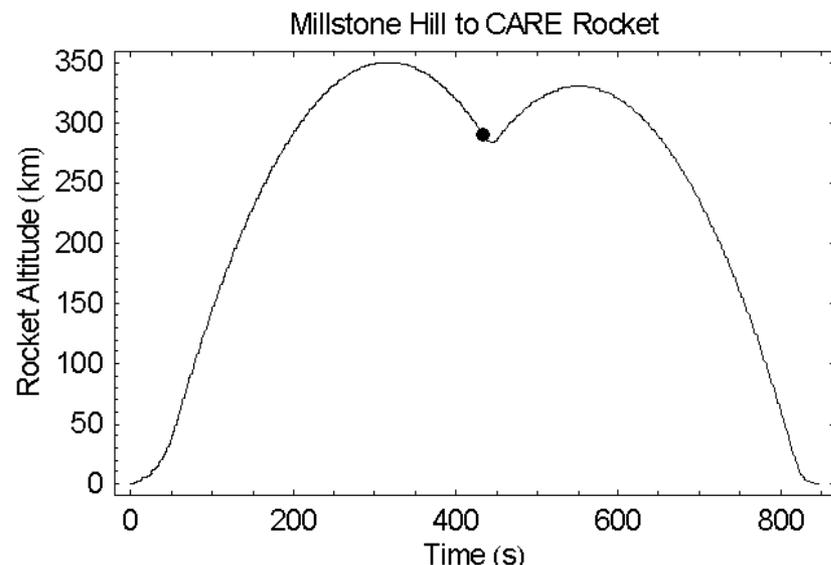
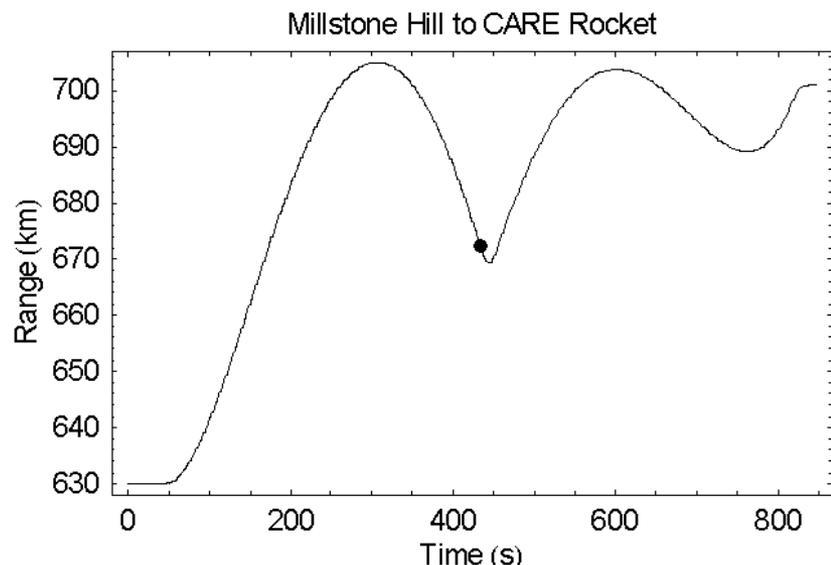
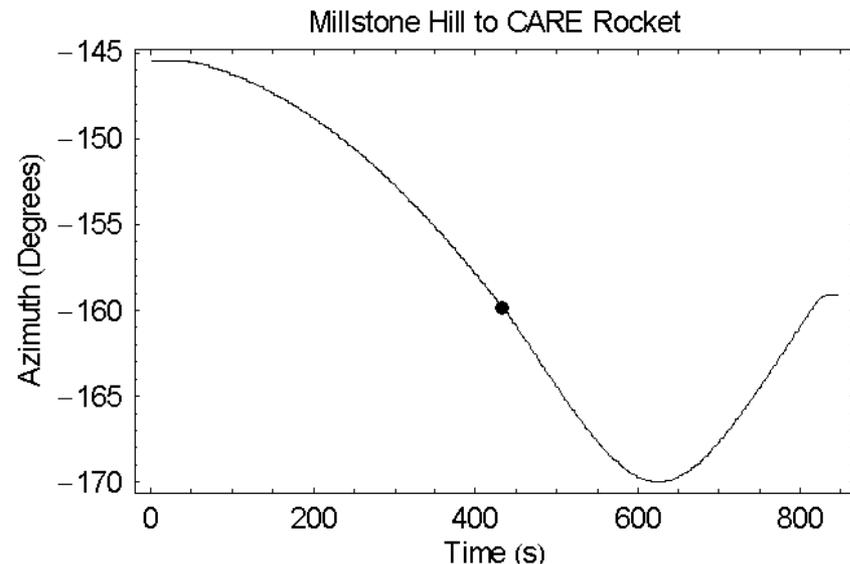
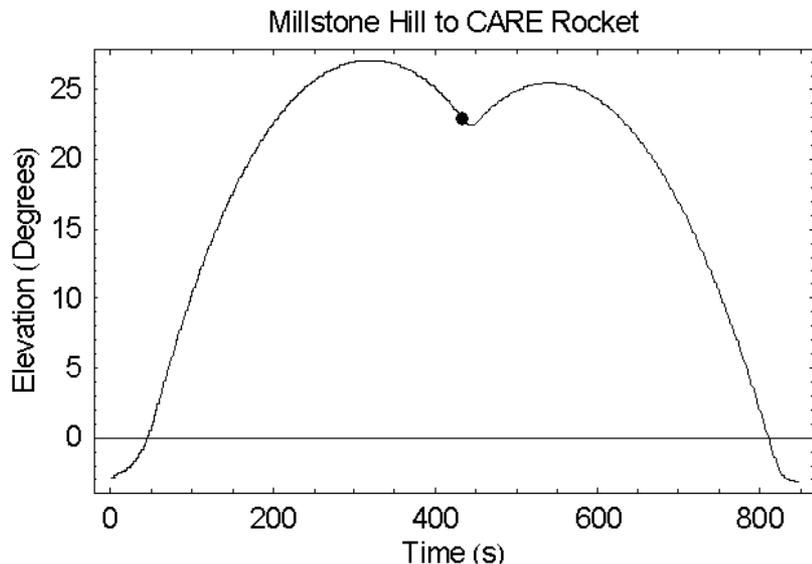


CARE Rocket Trajectory





CARE Chemical Release Position Relative to the Millstone Hill UHF Radar for Nihka with Nozzle



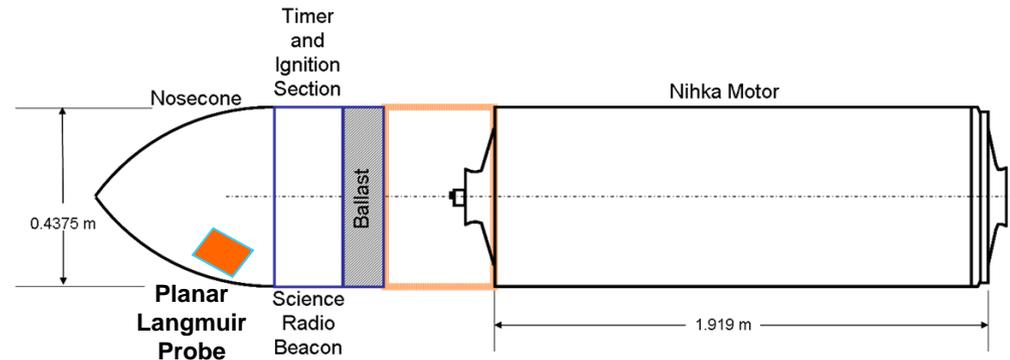
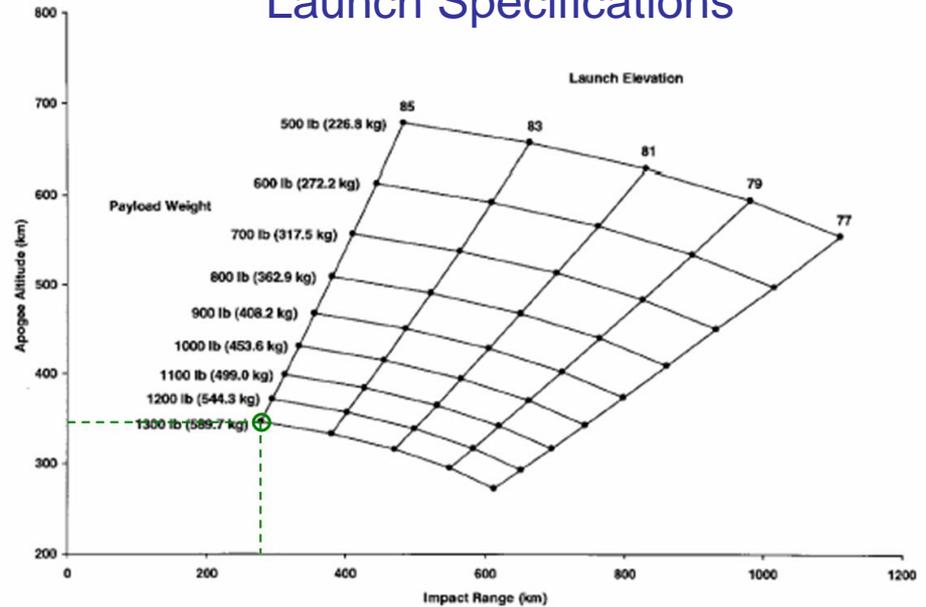


Single Nihka Motor Payload



- Nihka
 - Gross Mass: 399 kg (879 lb), Empty Mass: 70 kg (154 lb).
 - Propellants: Solid.
 - Thrust (vac): 50.500 kN (11,353 lbf).
 - Isp: 285 sec., Burn time: 17 sec.
- Total Chemical Payload Mass
 - Gross Nihka Mass 400 kg
 - Supporting Hardware 190 kg
 - Total Mass 590 kg
- Payload Apogee 350 km
- CARE Release Point ~280 km on Downleg
 - Chemical Payload Descending at 1 km/s
 - Exhaust Injection at 2 km/s Downward
 - Full Dust Cloud Released in 17 Seconds

Launch Specifications



Payload Components

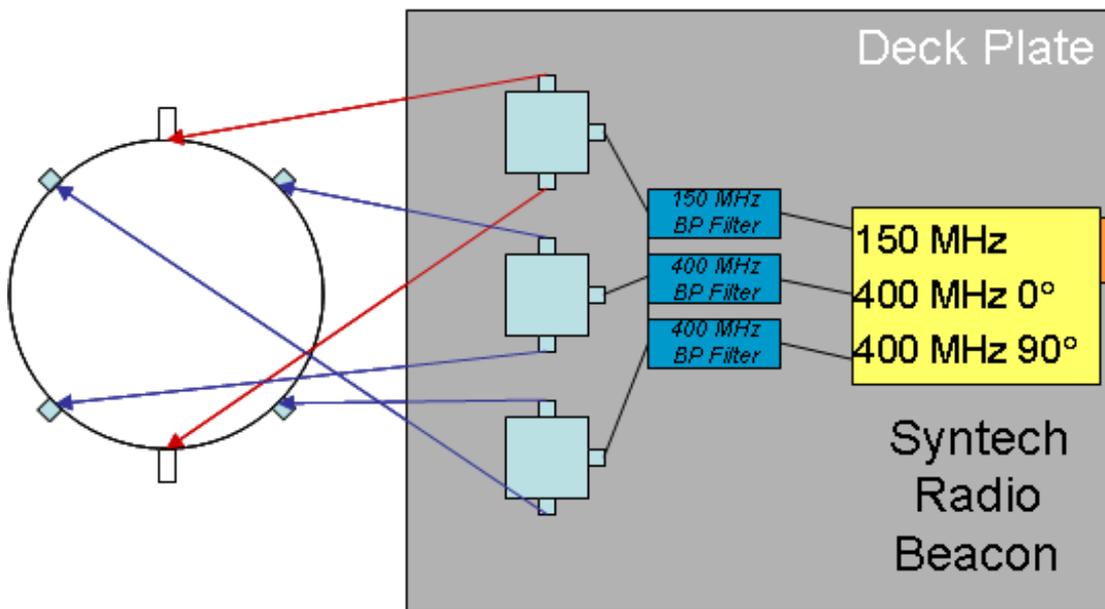
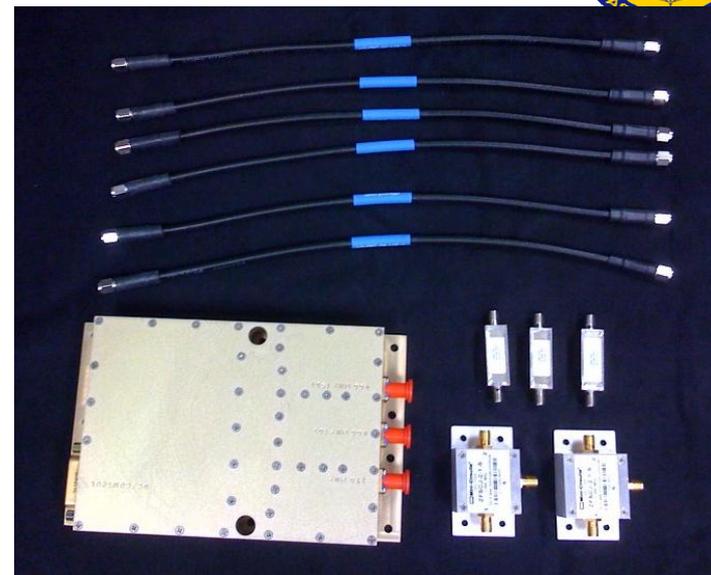


NRL CERTO Rocket Beacon

Paul Bernhardt, Matt Wilkens



- Plasma Measurements
 - Radio Beacon (Bernhardt)
 - Radio Scintillations at VHF and UHF
 - Background Electron Density
 - Electron Removal by Chemical Release



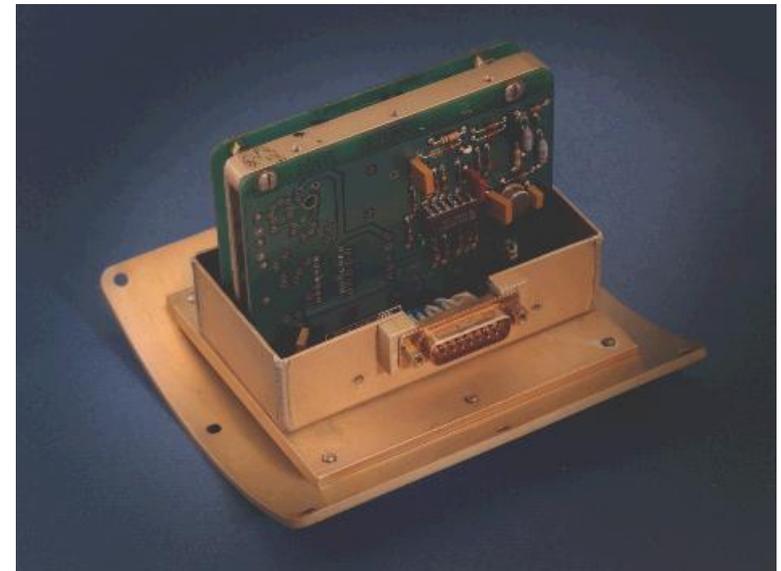
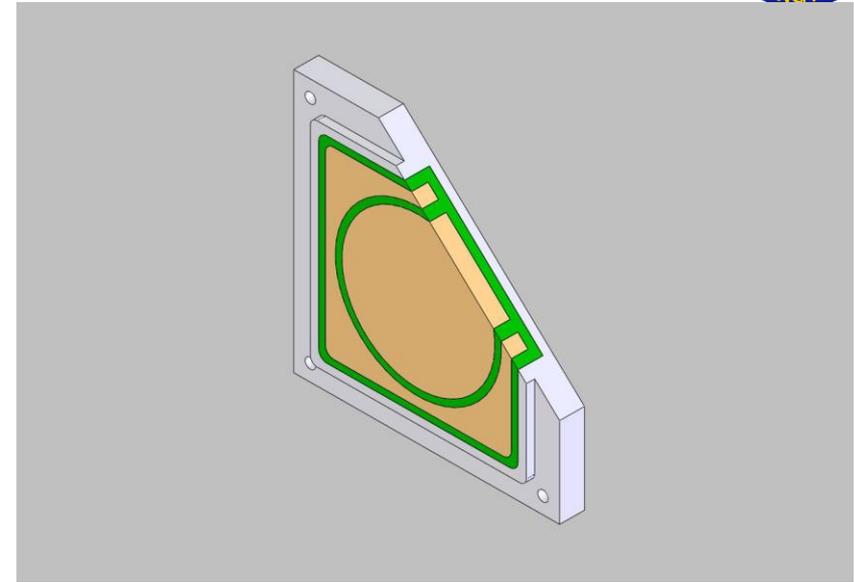


AFRL Planar Langmuir Probe

John Balenthin, PI

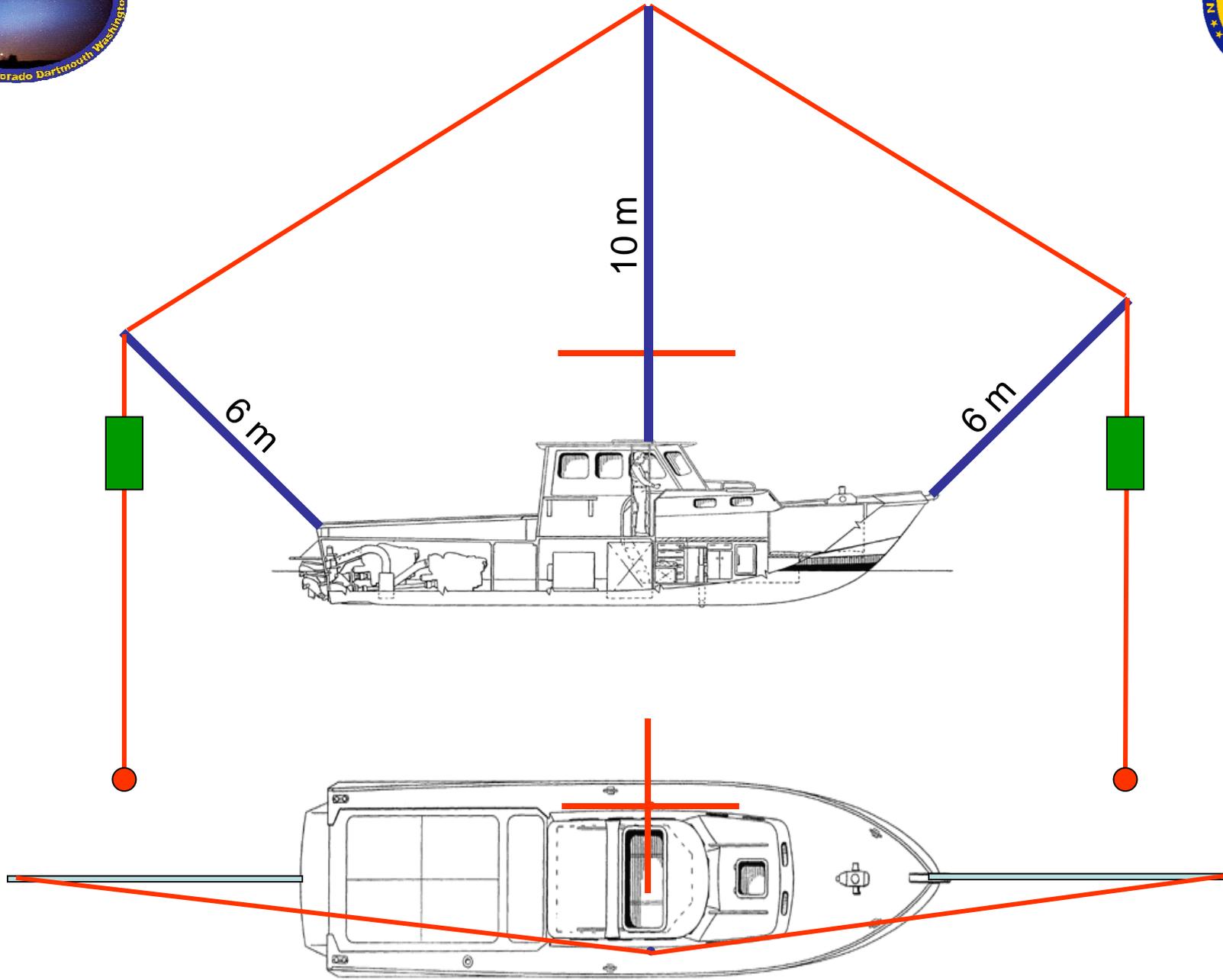


- Ion Density at Nihka Motor Section
- Backplate
 - 1/2", 5"x5" Square Aluminum
 - 4"x4" Raised Portion Flush to Skin.
- Behind Probe
 - 1/8" thick aluminum
 - Teflon (1/8" flat plate) for High Temperature
 - G10 epoxy board for Low Temperature
- Gold Conductor
 - Square 2.3" x 2.3"
 - Circle 1.5" Diameter
 - 1/4" thick
- Gold Plated Stainless Steel
 - 304 grade
 - Attached by Screws Through Back Panel
- Connector Pins
 - Pin 1: Ch1+ 6000s/s A/D channel
 - Pin 9: Twisted pair or shield ground for Pin1
 - Pin 2: Ch1- 1000s/s A/D channel
 - Pin 10: Twisted pair or shield ground for Pin2
 - Pin 7: +28VDC
 - Pin 14: 28V Return
 - Pin 8: +28VDC
 - Pin 15: 28V Return



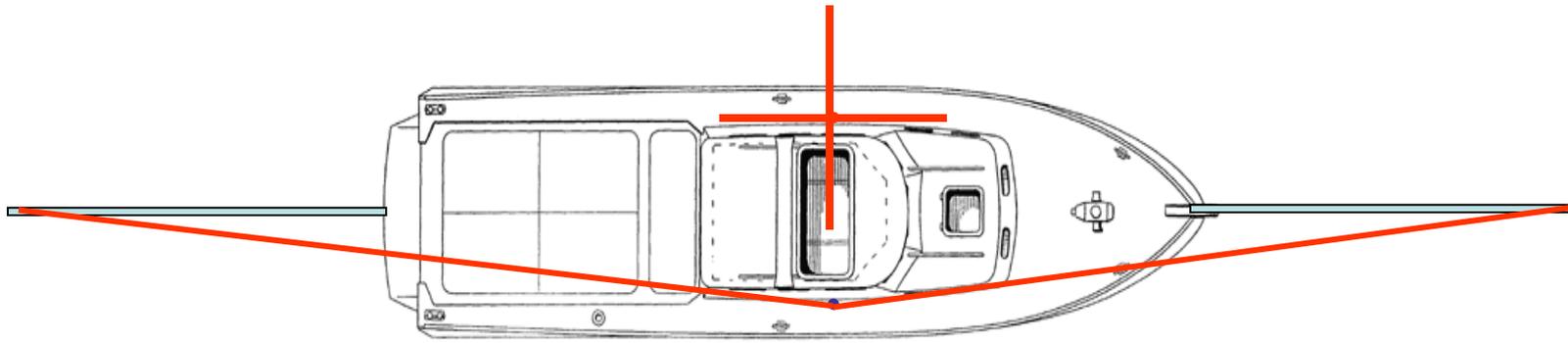


Instrumented Thomas Reed Boat for CARE



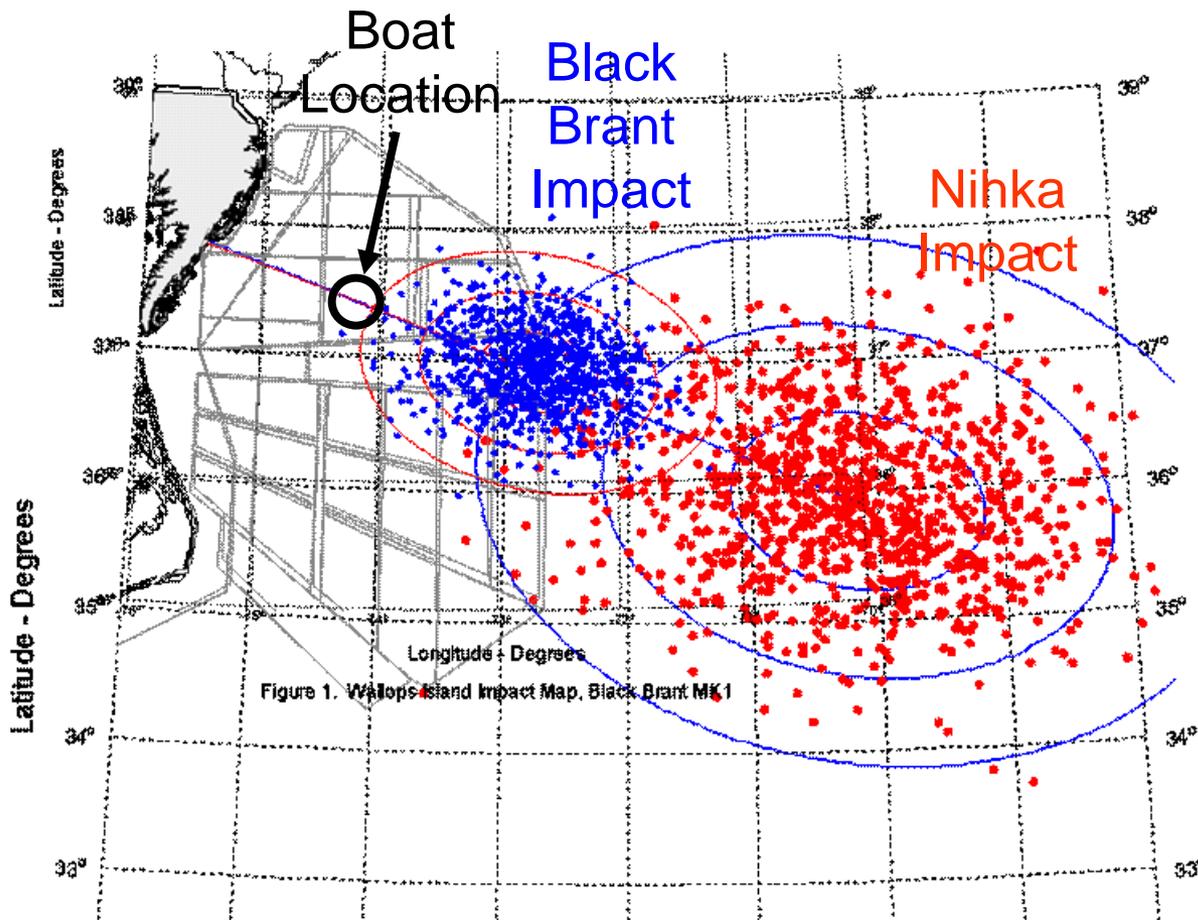


Instrumented Thomas Reed Boat for CARE





Instrumented Boat Station





Satellite Sensor for CARE



- Diagnostics of CARE Artificial Dust Cloud
 - AIM Panoramic UV Nadir Imaging System
- Global Transport of Dust Layer





CARE Launch Window



- Requirements
 - Clear Skies at 3 Observation Sites
 - Moon Below Horizon
 - Evening Sun at 12 to 18 Degrees Depression Angle
 - No Sporadic-E One Hour Before Launch
 - Sporadic-E Will Interfere with Dust Radar Scatter Observations
- Launch Window
 - Release Time
 - Date 8-19 September 2009
 - Release Time 19:18 to 19:00 EST (Date Dependent)
 - Optical Observation Period
 - Date 8-19 September 2009
 - Viewing Until 19:50 to 19:31 EST (Date Dependent)
 - Viewing Time ~ 30 Minutes
 - Sporadic-E is Bad in August so September Chosen